## **AMENDMENTS TO THE CLAIMS:**

Please cancel without prejudice claim 24 and amend claims 1, 2 and 22 as follows.

This listing of claims will replace all prior versions, and listings, of claims in the application:

1. (currently amended) A method of assembling components together in sealed relationship, the components have respective mating surfaces, the method including the steps of:

applying to <u>aat least one</u> mating surface a layer of polysulphide sealant and allowing the sealant to cure;

after allowing said sealant to cure, bringing together the mating surfaces; and applying a pre-determined pressure therebetween for a pre-determined period whereby to bring about a sealed joint between the two mating surfaces.

- 2. (currently amended) A method as in claim 1 in which a-said layer of polysulphide sealant is applied to both mating surfaces.
- 3. (previously presented) A method as in claim 1 in which the period of application of pressure is at least 1 hour.
- 4. (original) A method as in claim 3 in which the said period is between 5 and 1 x  $10^7$  hours.

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- 5. (original) A method as in claim 3 in which the said period is between 8 and 1440 hours.
- 6. (previously presented) A method as in claim 1 in which the pre-determined pressure is between 5 and 400 MPa.
- 7. (previously presented) A method as in claim 1 in which the pre-determined pressure is between 5 and 200 MPa.
- 8. (previously presented) A method as in claim 1 in which the pre-determined pressure is between 8 and 50 MPa.
- 9. (previously presented) A method as in claim 1 in which the pre-determined pressure is applied by bolting together the two components in their final assembled configuration.
- 10. (previously presented) A method as in claim 1 in which the components are subject to a raised temperature during at least part of the step of applying pressure.
- 11. (previously presented) A method as in claim 1 in which the at least one layer of polysulphide sealant is applied to a painted said mating surface.
- 12. (original) A method as in claim 11 in which the layer of polysulphide sealant is applied to the painted mating surface a sufficiently short time after the paint is applied to at least

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substantially reduce the need for any further treatment of the painted surface prior to the application of the layer of polysulphide sealant.

13. (original) A method as in claim 12 in which the layer of polysulphide sealant is applied to the painted surface immediately after the paint has dried.

14. (original) A method as in any preceding claim in which the components with sealant applied are stored, including the step of applying a protective covering to the cured layer of polysulphide sealant prior to storage of the component.

15. (previously presented) A method as in claim 1, in which the mating surface to which the layer of polysulphide sealant is not applied is a painted surface.

16. (previously presented) A method as in claim 1 in which the layer of polysulphide sealant applied is a transition metal oxide curing compound.

17. (previously presented) A method as in claim 1 in which the layer of polysulphide sealant applied is a manganese dioxide curing compound.

18. (previously presented) A method as in claim 1 in which the layer of polysulphide sealant applied is a dichromate curing compound.

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19. (previously presented) A method as in claim 1 in which the layer of polysulphide

sealant applied is an organic-cure compound.

20. (previously presented) A method of assembling components together as in claim 1 in

which the components comprise aircraft structural components.

21. (original) A method as in claim 20 in which the aircraft structural components are

used to house fuel on board the aircraft.

22. (currently amended) An assembly of two components for forming a fluid-tight seal

together, each component having a mating surface for sealing to a mating surface of the other

component, said assembly comprising at least one said mating surface having a layer of eured

polysulphide sealant <u>cured</u> thereon <u>prior to assembly</u>.

23. (original) An assembly as in claim 22 in which the components will form part of a

fuel storage system for an aircraft.

24. (cancelled).

25. (cancelled).

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